

Integrated simulation over aircraft, rotorcraft and engines

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● Responsible Representative

Atsushi Hashimoto, Aviation Technology Directorate, Aircraft Lifecycle Innovation Hub

● Contact Information

Kanako Yasue(yasue.kanako@jaxa.jp)

● Members

Hirokazu Higashida, Manabu Hisida, Atsushi Hashimoto, Mami Hayakawa, Kenji Hayashi, Tomoaki Ikeda, Takashi Ishida, Masatoshi Kanayama, Ryohei Kirihara, Masashi Kanamori, Takuhito Kuwabara, Yuichi Matsuo, Shingo Matsuyama, Yasuhiro Mizobuchi, Taisuke Nambu, Hideji Saiki, Kei Shimura, Atsushi Shinozuka, Andrea Sansica, Shota Taniguchi, Kanako Yasue, Hiroki Yao

● Abstract

An aircraft is a large-scale complex system consisting of mechanics, electronics, and software. As a result, the trend of increasing development time and costs is unstoppable. Model-Based Systems Engineering (MBSE) is attracting attention as a technology to solve this problem. The objective of this research is to develop advanced MBSE technology that combines JAXA's strengths in numerical simulation technology and AI technology in order to reduce the cost of aircraft development. JSS3 will be used to develop multidisciplinary integrated simulation technology, including aerodynamics and structural analysis of passenger aircraft, aerodynamics and noise analysis of rotorcraft (eVTOL), and aerodynamics and combustion analysis of aircraft engines.

Ref. URL: <https://www.aero.jaxa.jp/eng/research/basic/numerical/>

● Reasons and benefits of using JAXA Supercomputer System

JSS is necessary to complete large scale numerical simulations of unsteady phenomena and to understand it in short time span.

● Achievements of the Year

We performed the world's first global stability analysis of buffett at flight Reynolds numbers. We predicted buffet onsets and found buffet cells (Fig. 1a) and side-of-body separation (Fig. 1b) modes. The side-of-body separation mode was found for the first time. This work received the 40th ANSS Grand Prize (Numerical Simulation Technology Division).

FaSTAR-Move have been extended to enable noise prediction analysis by coupling with rNoise which can simulate rotorcraft noise utilizing FW-H method. Noise prediction for a rotor with 4 blades was performed and

reasonable noise carpet was obtained (Fig. 2).

A noise propagation analysis tool, iAESOME, has been stabilized and accelerated by improving the matrix calculation. We achieved large-scale analysis of 3 billion points within 20 hours using JSS3 (Fig.3).

Functions for LES combustion analysis have been implemented in HINOCA-AE. Comparison of outlet temperatures with combustion tests shows that the error with the measured values is within 100K (Fig.4).

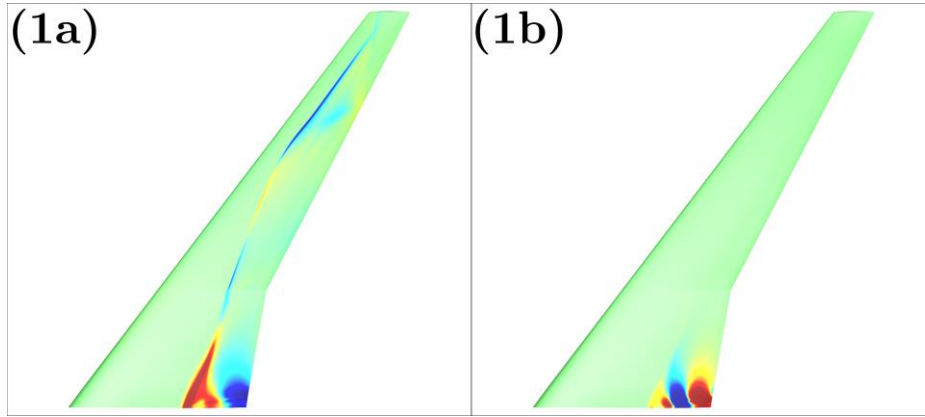


Fig. 1: Flight Reynolds number buffet cells (Fig. 1a) and side-of-body separation (Fig. 1b) modes.

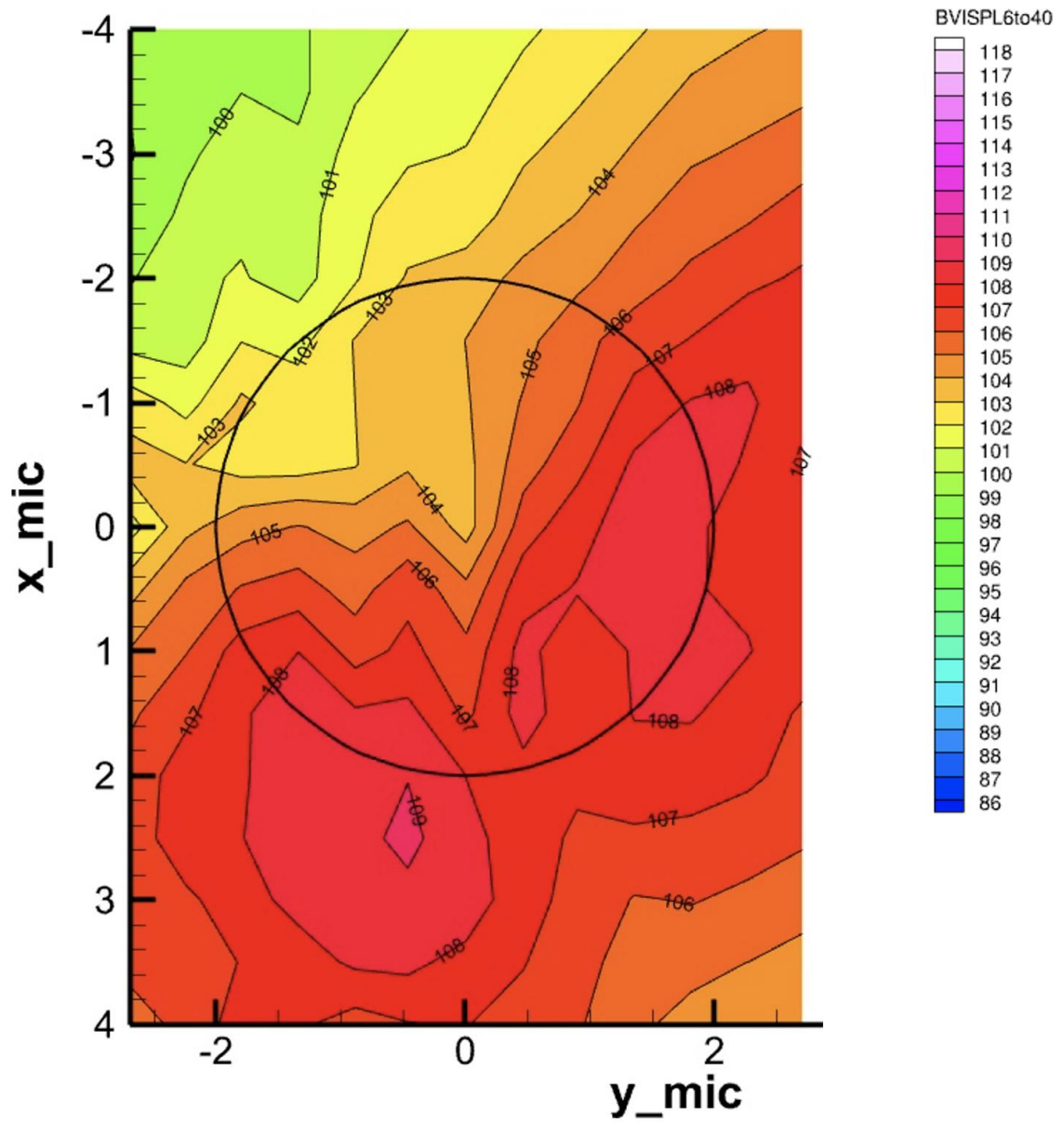


Fig. 2: Obtained noise carpet of rotating blades.

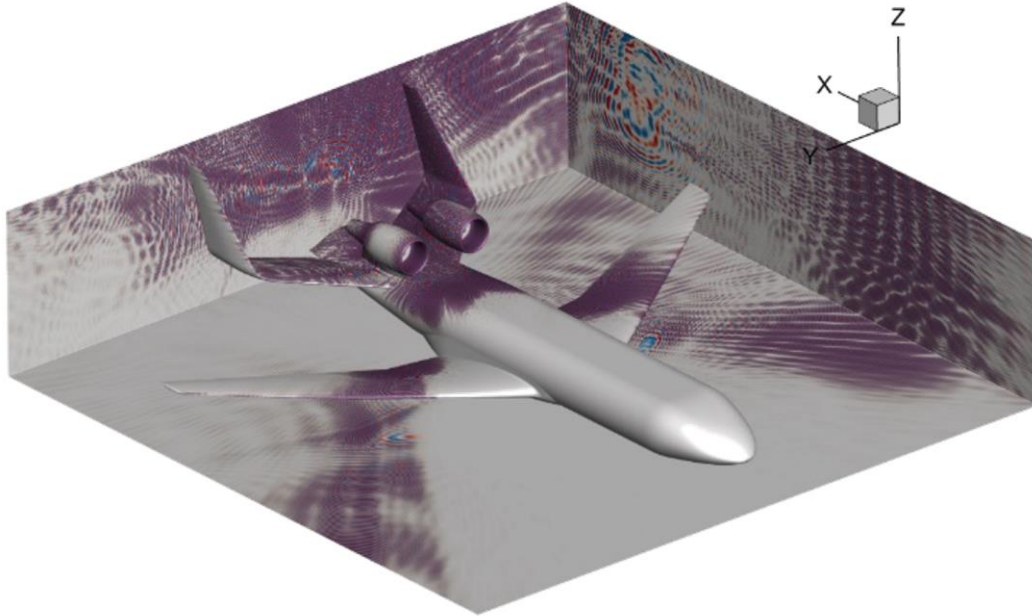


Fig. 3: Noise shielding effect analysis over a aircraft with engines.

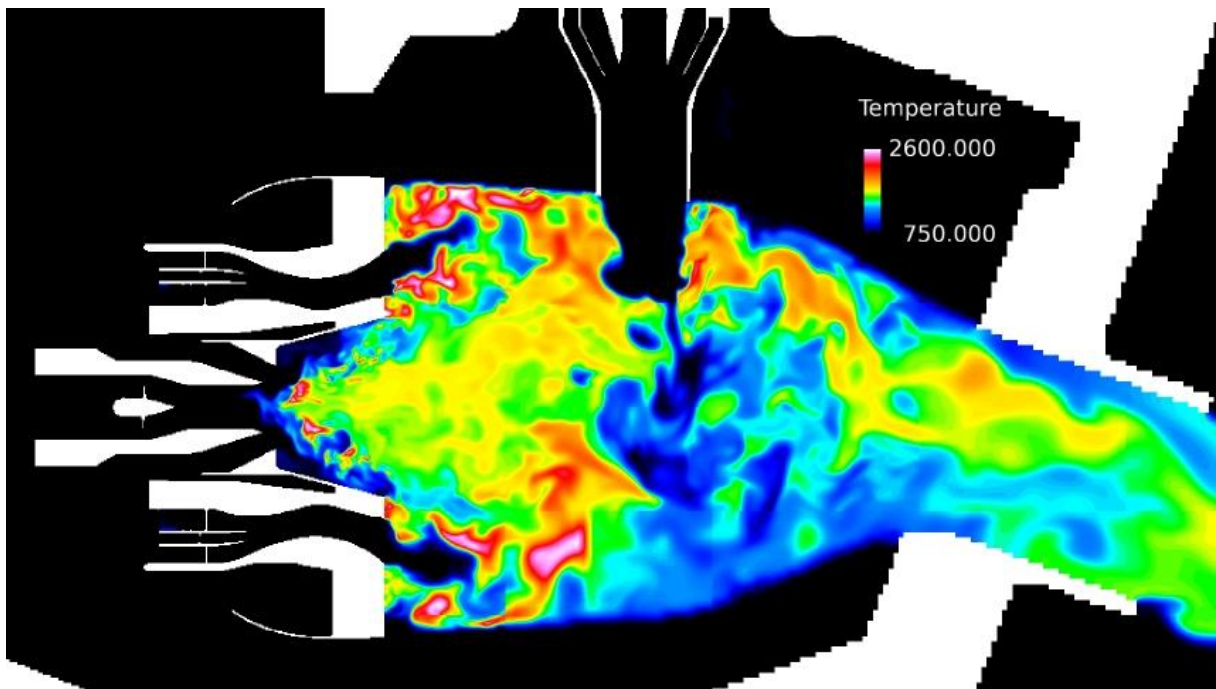


Fig. 4: A temperature distribution of combustor analyzed by HINOCA-AE

● Publications

- Peer-reviewed papers

[1] Sansica, A. and Hashimoto, A., "Global Stability on Large Three-Dimensional Grids for Full-Aircraft Transonic Buffet Characterization at Flight Reynolds Numbers", AIAA Journal (under review)

- Non peer-reviewed papers

[1] Sansica, A. and Hashimoto, A., "Turbulent transonic buffet onset prediction on the NASA Common Research Model via global stability analysis", vol. JAXA-SP-20-008, ISSNONLINE 2433-2232, pp. 109-118 (available at <https://jaxa.repo.nii.ac.jp>), 2022.

[2] Sartor, F., Sansica, A., Hayashi, K., Yamamoto, T., Ishida, T. and Hashimoto, A., "Assessment of URANS and ZDES Simulations for Turbulent Transonic Buffet Predictions on Full-Aircraft Configurations", vol. JAXA-SP-22-007, ISSNONLINE 2433-2232, pp. 341-348 (available at <https://jaxa.repo.nii.ac.jp>), 2022.

[3] Sansica, A. and Hashimoto, A., "Turbulent Transonic Buffet Prediction on the NASA Common Research Model via Global Stability Analysis: From Wind Tunnel to Flight Reynolds Numbers", AIAA Scitech Forum and Exposition, National Harbor, MD, January 2023.

- Oral Presentations

[1] Sansica, A. and Hashimoto, A., "Turbulent transonic buffet onset prediction on the NASA Common Research Model via global stability analysis", 54th Fluid Dynamics Conference / the 40th Aerospace Numerical Simulation Symposium, Morioka, Japan, June 2022

[2] Sartor, F., Sansica, A., Hayashi, K., Yamamoto, T., Ishida, T. and Hashimoto, A., "Assessment of URANS and ZDES Simulations for Turbulent Transonic Buffet Predictions on Full-Aircraft Configurations", 54th Fluid Dynamics Conference / the 40th Aerospace Numerical Simulation Symposium, Morioka, Japan, June 2022

[3] Sansica, A. and Hashimoto, A., "Turbulent transonic buffet onset prediction on the NASA Common Research Model via global stability analysis", 12th International Symposium on Turbulence and Shear Flow Phenomena (TSFP12), Osaka, Japan (Online), July 2022

[4] Sansica, A. and Hashimoto, A., "Turbulent Transonic Buffet Prediction on the NASA Common Research Model via Global Stability Analysis: From Wind Tunnel to Flight Reynolds Numbers", AIAA Scitech Forum and Exposition, National Harbor, MD, January 2023. View video presentation: <https://doi.org/10.2514/6.2023-1989>

● Usage of JSS

● Computational Information

Process Parallelization Methods	MPI
Thread Parallelization Methods	N/A
Number of Processes	480 - 8472
Elapsed Time per Case	200 Hour(s)

● **JSS3 Resources Used**

Fraction of Usage in Total Resources*1(%): 1.36

Details

Computational Resources		
System Name	CPU Resources Used (core x hours)	Fraction of Usage*2(%)
TOKI-SORA	36,843,633.27	1.61
TOKI-ST	24,437.70	0.02
TOKI-GP	0.00	0.00
TOKI-XM	51.19	0.03
TOKI-LM	8,239.76	0.55
TOKI-TST	1,349.33	0.04
TOKI-TGP	0.00	0.00
TOKI-TLM	0.00	0.00

File System Resources		
File System Name	Storage Assigned (GiB)	Fraction of Usage*2 (%)
/home	1,656.01	1.50
/data and /data2	166,616.88	1.29
/ssd	35,383.39	4.90

Archiver Resources		
Archiver Name	Storage Used (TiB)	Fraction of Usage*2 (%)
J-SPACE	14.56	0.06

*1: Fraction of Usage in Total Resources: Weighted average of three resource types (Computing, File System, and Archiver).

*2: Fraction of Usage : Percentage of usage relative to each resource used in one year.

● **ISV Software Licenses Used**

ISV Software Licenses Resources		
	ISV Software Licenses Used (Hours)	Fraction of Usage ^{*2} (%)
ISV Software Licenses (Total)	377.96	0.26

*2: Fraction of Usage : Percentage of usage relative to each resource used in one year.